

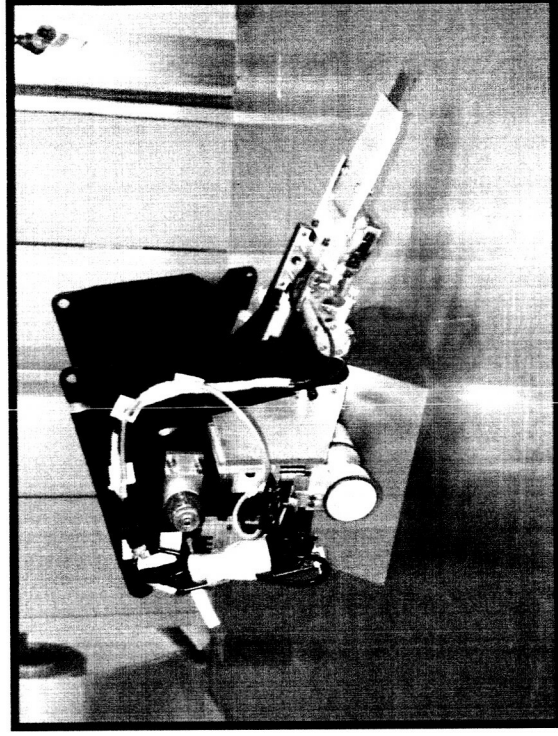


# Marshall Space Flight Center

Engineering Directorate  
Materials, Processes and Manufacturing Department



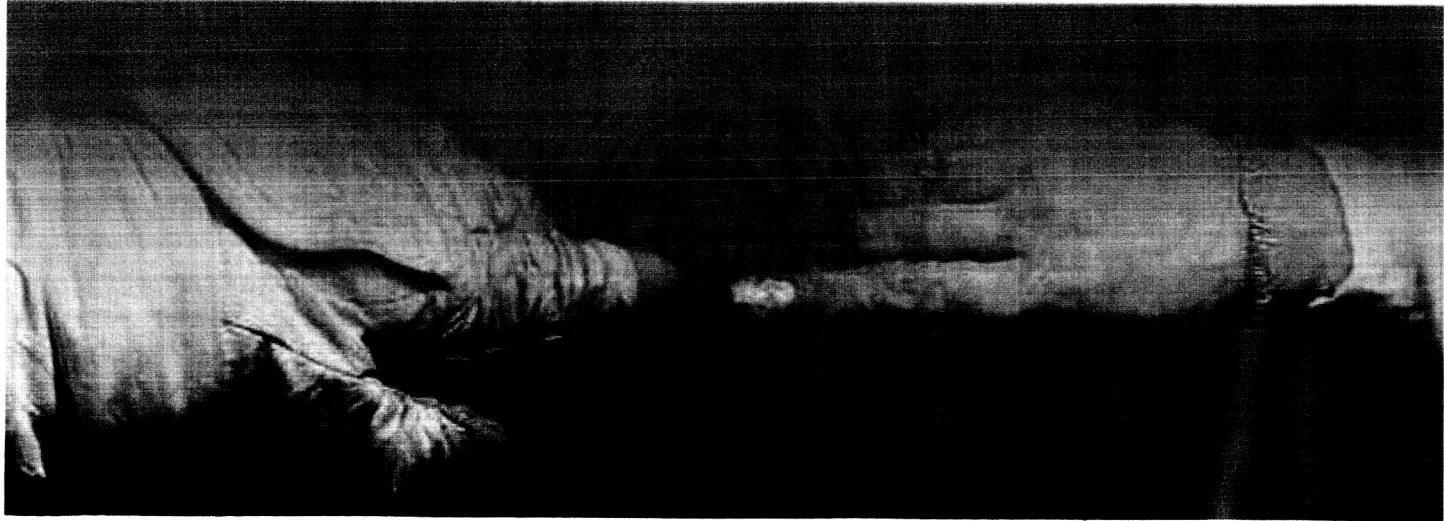
## Film Processing Module for Automated Fiber Placement



**Bruce Hulcher**  
NASA Marshall Space Flight Center

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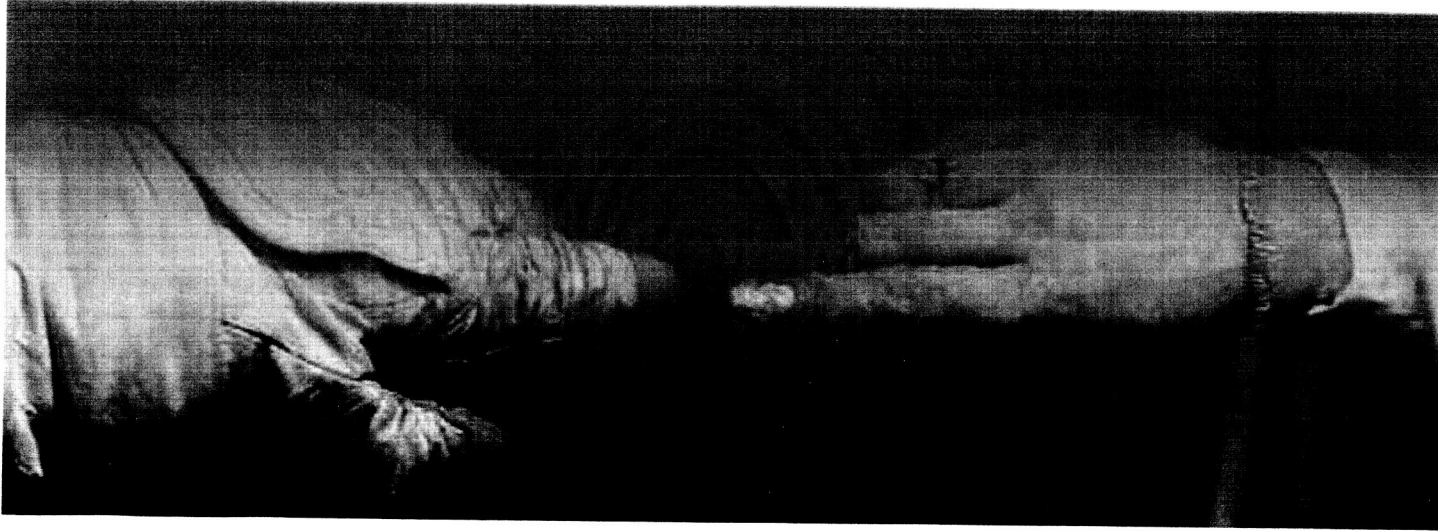
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#### *Outline*

1. Introduction and Background
2. Fiber Placement Process Description
3. Technology Benefits
4. Technology Description
5. Technical Details
6. Technical Advantages
7. R&D Status
8. Remaining Work
9. NASA Plans/Options
10. NASA's Tech Transfer Program
11. Acknowledgements





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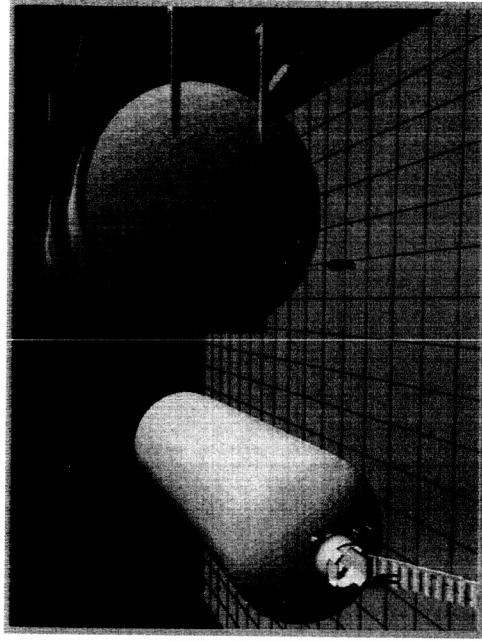
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### *Introduction and Background*

- Composite Processes and Fabrication Team, MPM Dept.
- Recent work focused on 2<sup>nd</sup> Gen RLV & NGLT Technology
- Micro-cracking/Permeability Issues for Liquid Cryogen Tanks
- Fabrication Trade Studies for Very Large Composite Tanks



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### *Introduction and Background: Genesis of Concept*

#### **Team Focus: *Reusability of Propellant Tanks***

- Micro-Cracking of Polymer Matrix Due to Thermal Cycling
- Manufacturability Issues: Out-of-Autoclave, Tooling Concepts, etc.
- Studies: Permeability of Various Polymers/Films to LOX, LH

#### **Not Captured:**

Fabrication of Structures Having Barrier Films/Foils and/or Core

Adhesives ⇨ Scale: Technology for Manufacturing of Large

(30 ft. x 90 ft.) Propellant Tanks



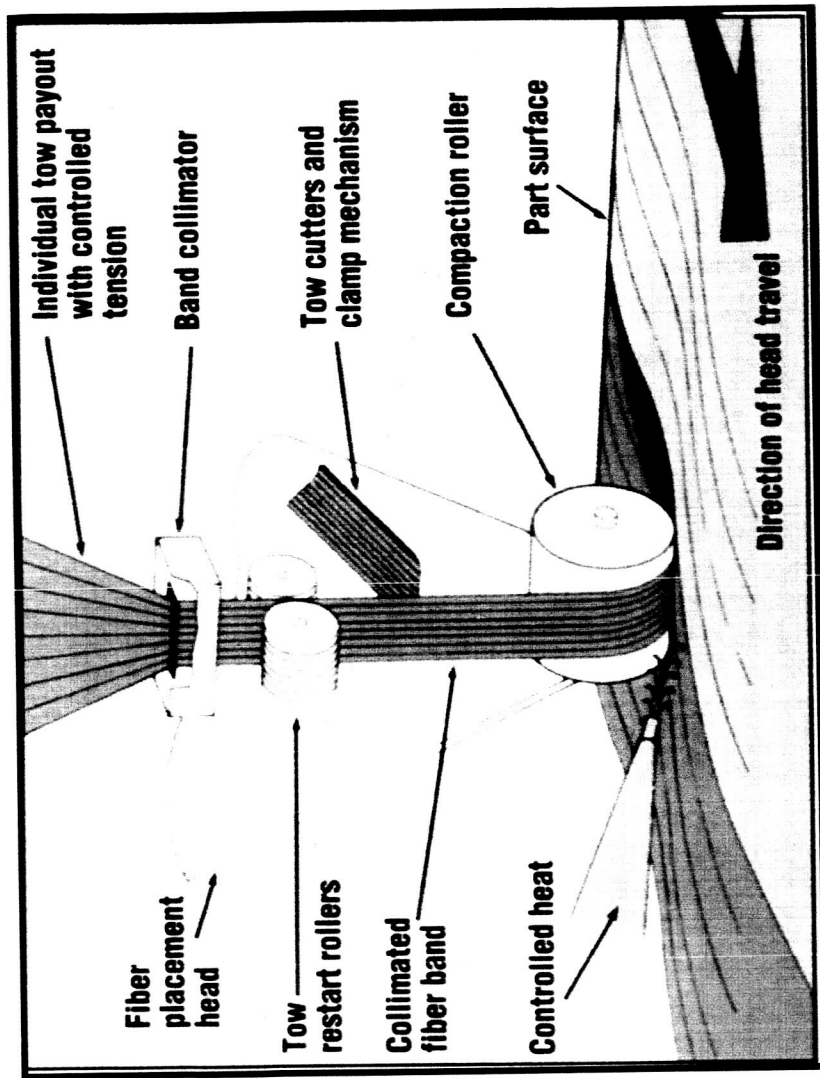
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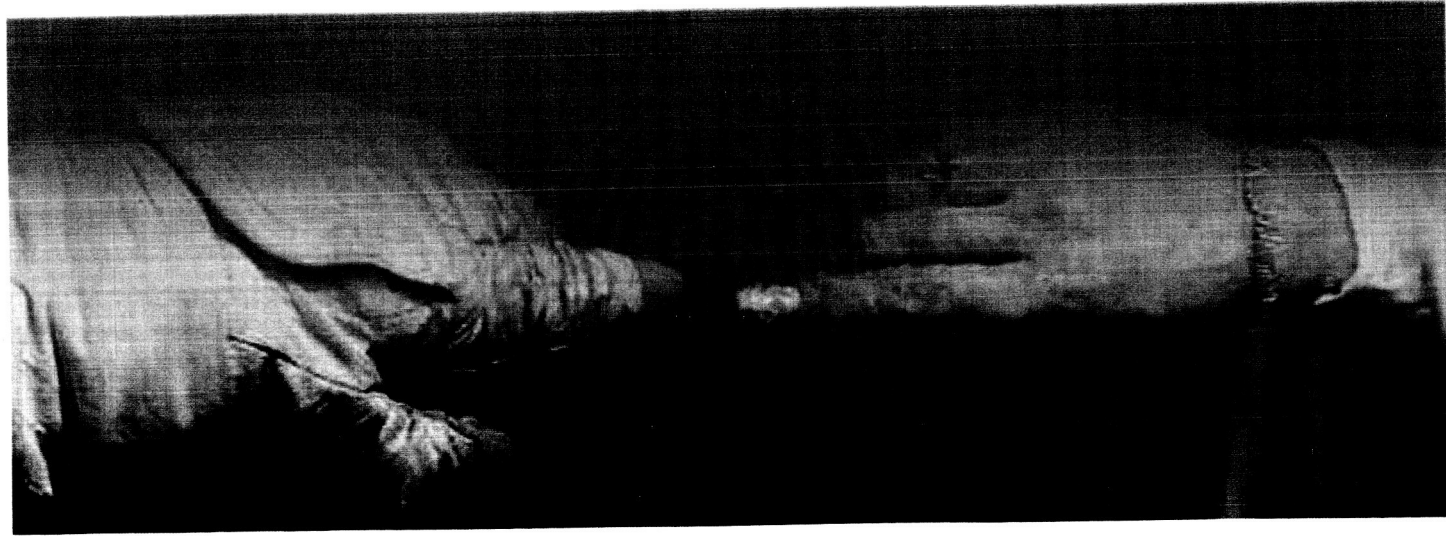


## *Fiber Placement Process Description*



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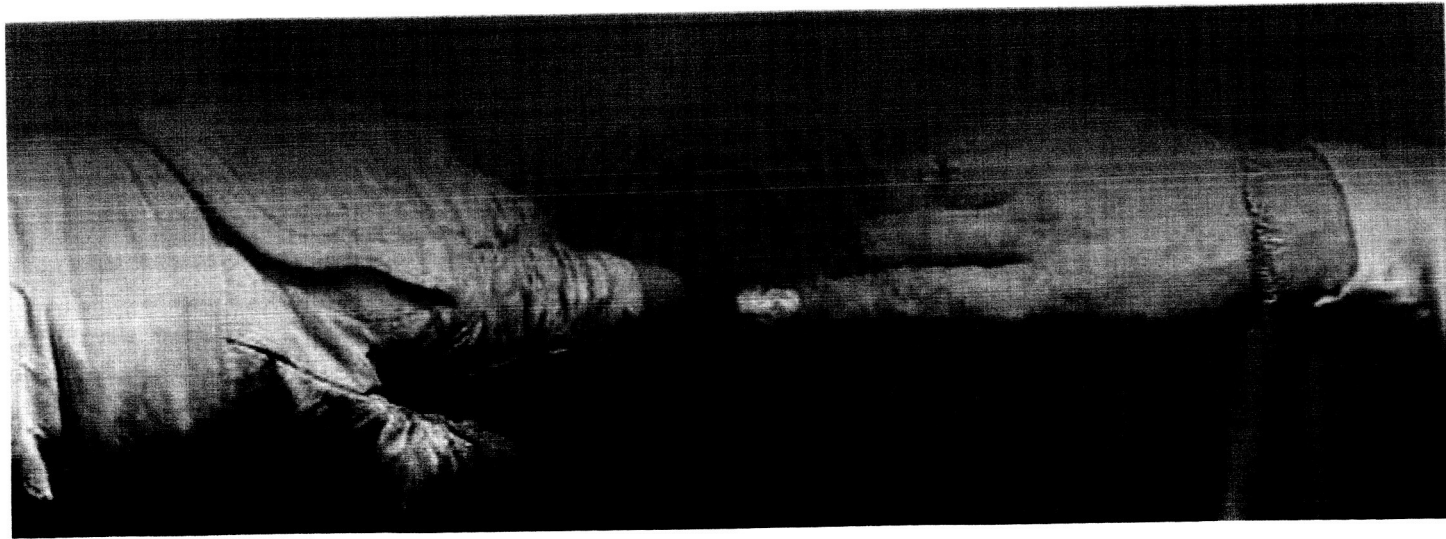


## *Technology Benefits*

- Enhancement of the Fiber Placement Process
- Simultaneous Placement of Films/Foils into Laminate During Ply Lay-up Cycle or as Separate Step
- Device May Be Designed as Add-On or Integrated into New Fiber Placement Machinery
- Ease of Attachment and Removal of Module to Host Machine
- Slave-Control Operation for Ease of Integration
- Reduction in Part Costs due to Reductions in Cycle Time
- Capability Extends to Automated Deposition of Core Adhesives

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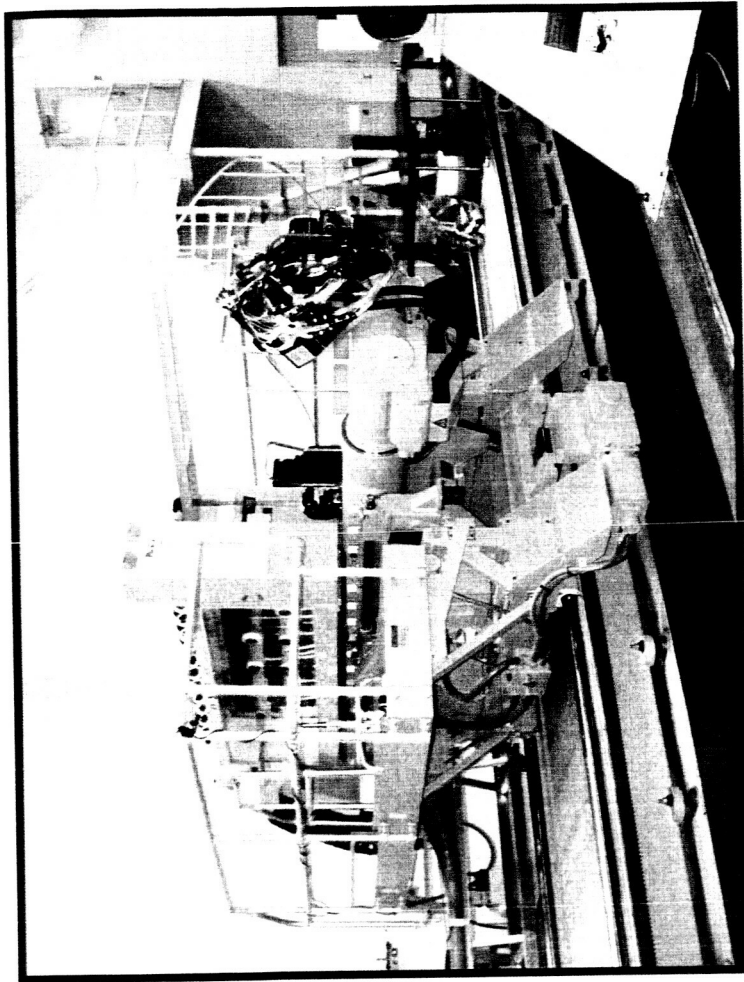


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*Technology Definition: Prototype Test Bed*

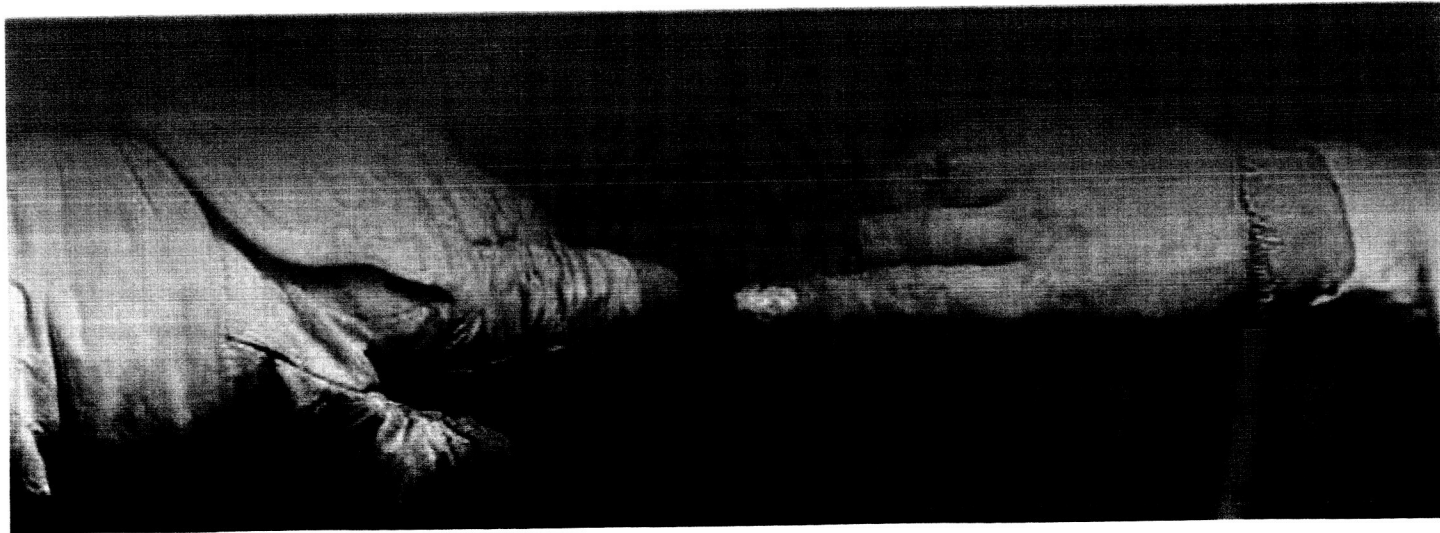


MSFC Fiber Placement Machine

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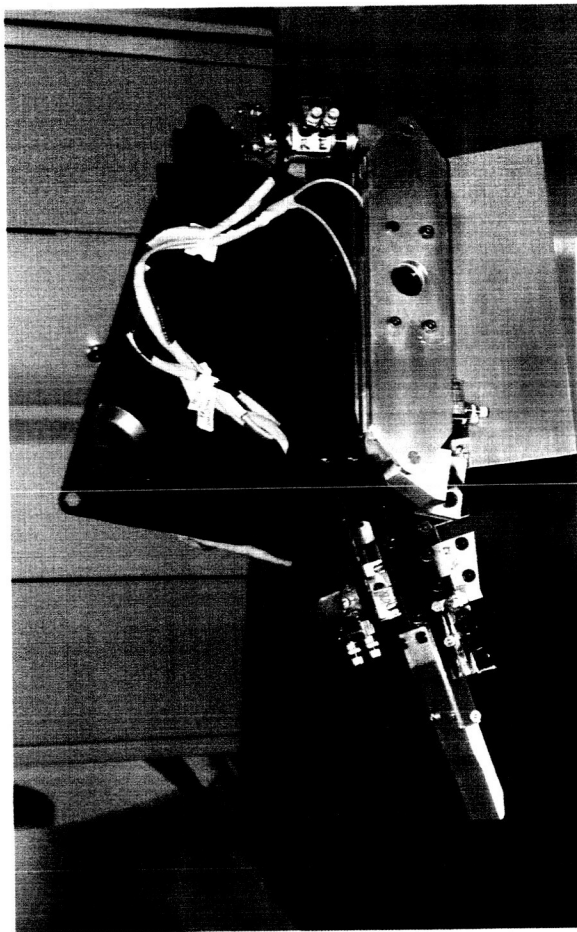
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## *Technology Definition: Module Specifications*



**Dimensions: 17in. x 8.5 in. x 10 in.**

**Weight: 12.5 lbs.**

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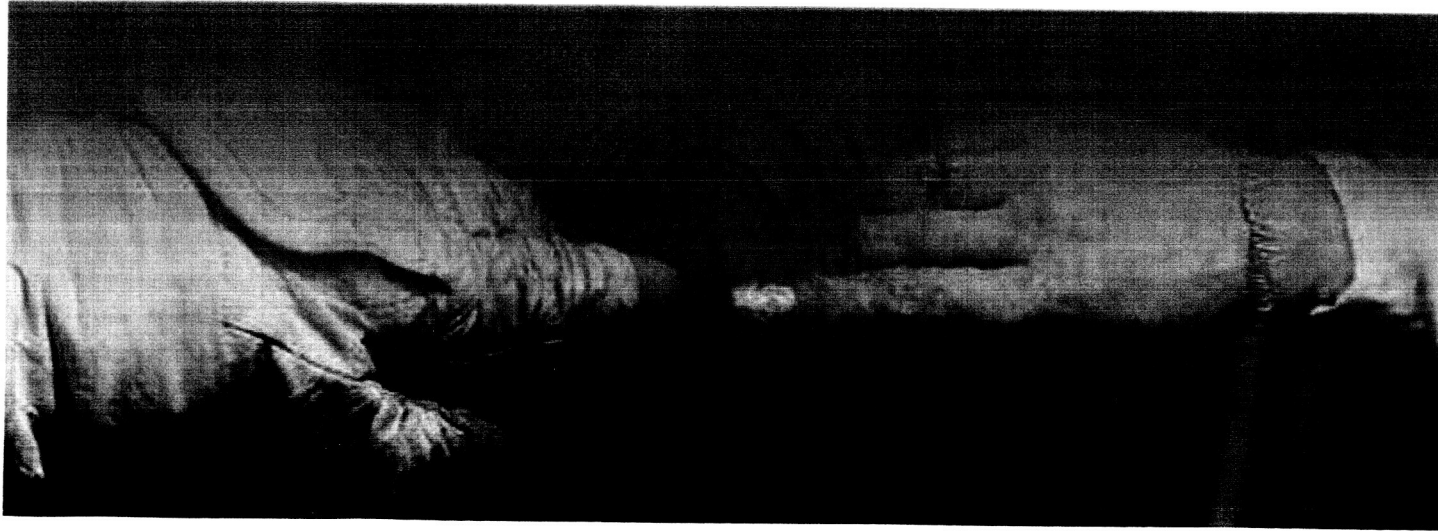


#### *Technical Details: Applications*

- Titanium/Graphite Laminates (TiGr)
- Glass/Epoxy/Aluminum Laminates (GLARE)
- Embedded Lightning Strike Protection
- Liners/Permeation Barriers
- Embedded Sensor Arrays
- Adhesives
- Processing

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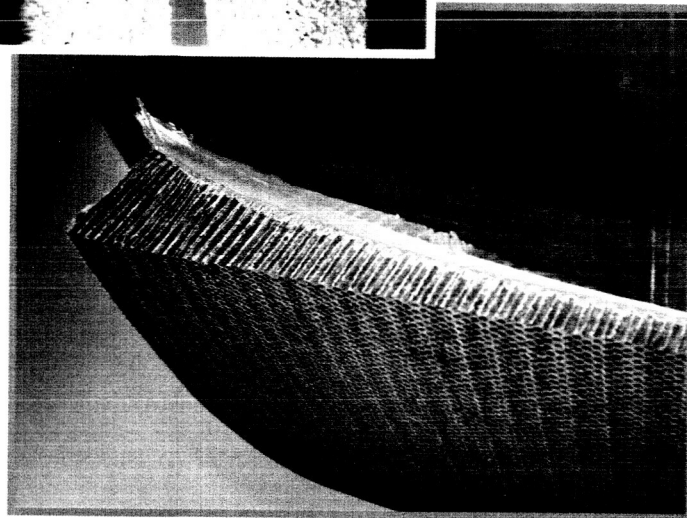
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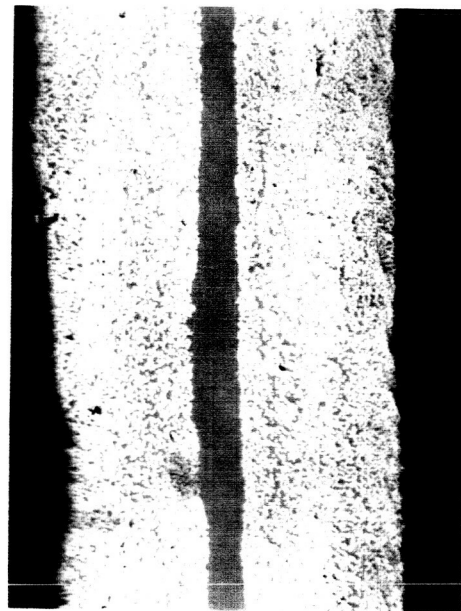
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## *Technical Details: Applications*



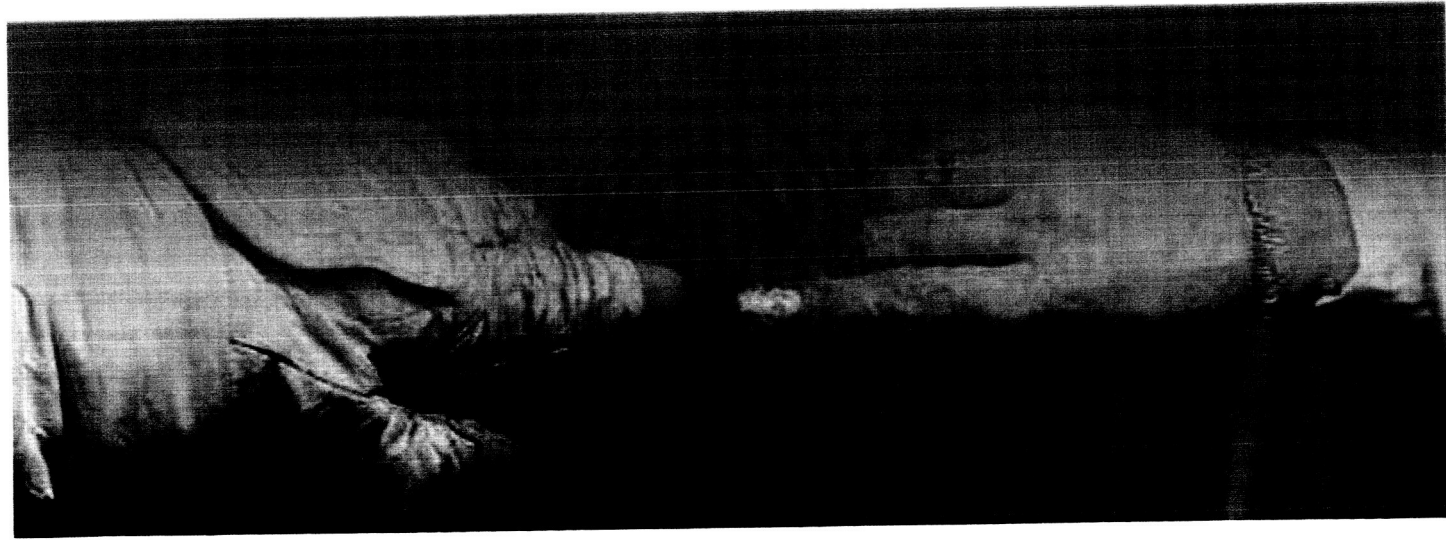
**Core Adhesives**



**Barrier Layers**

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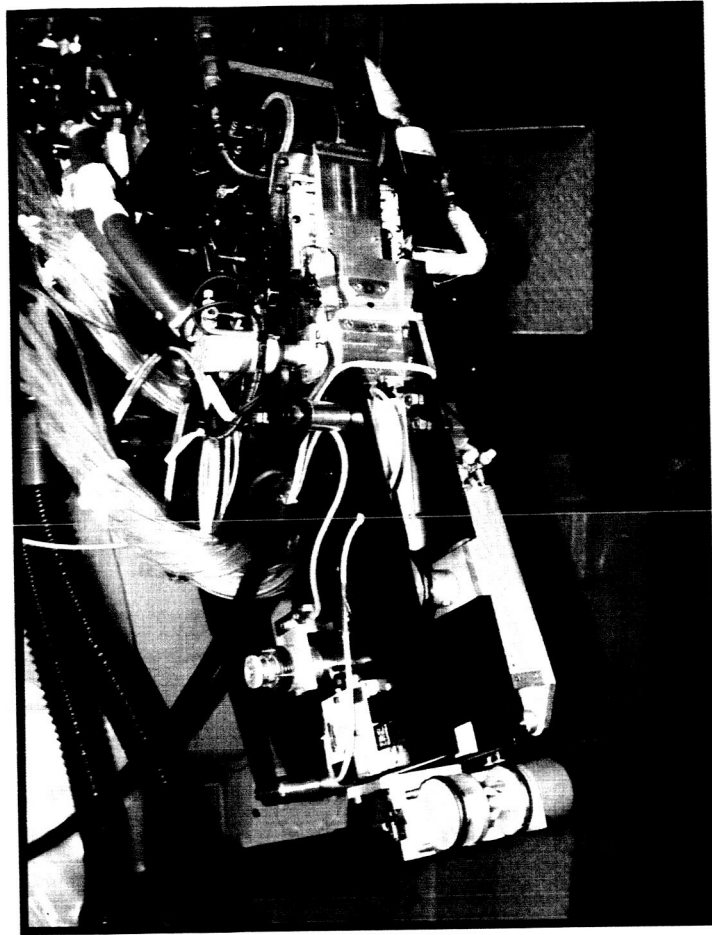
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*Technical Details: Attachment to FP Head*



**Film Module On Viper Placement Machine**

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*Technology Details: Simple Attachment to FPM*

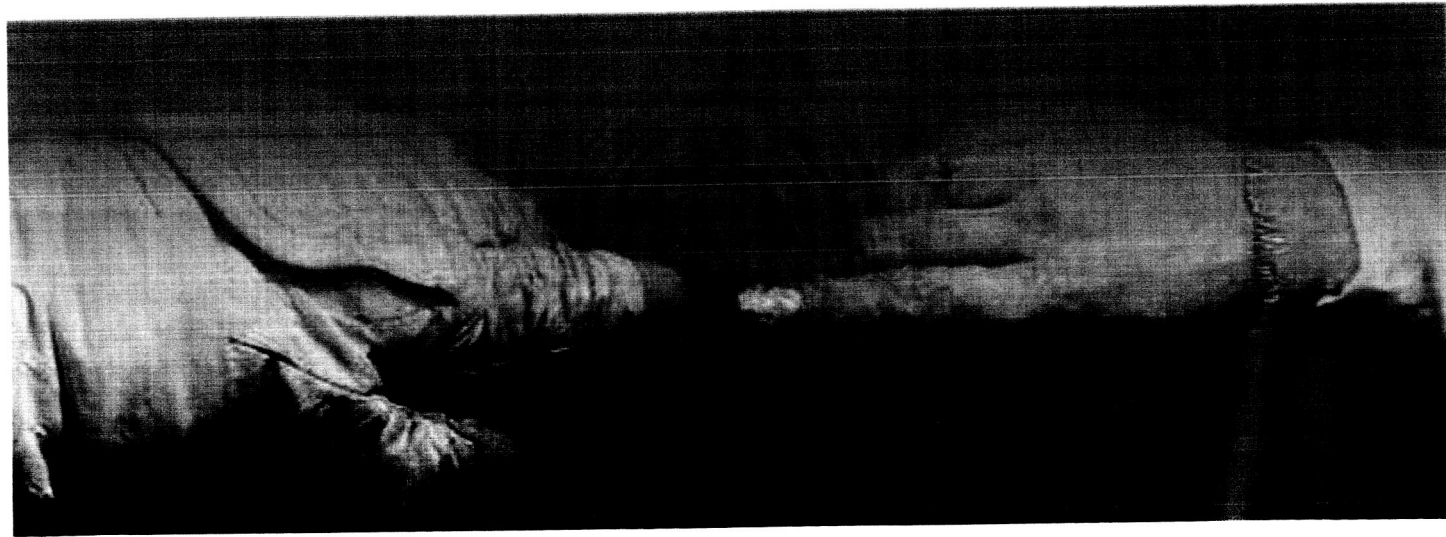


**Attachment Hardware for Module**

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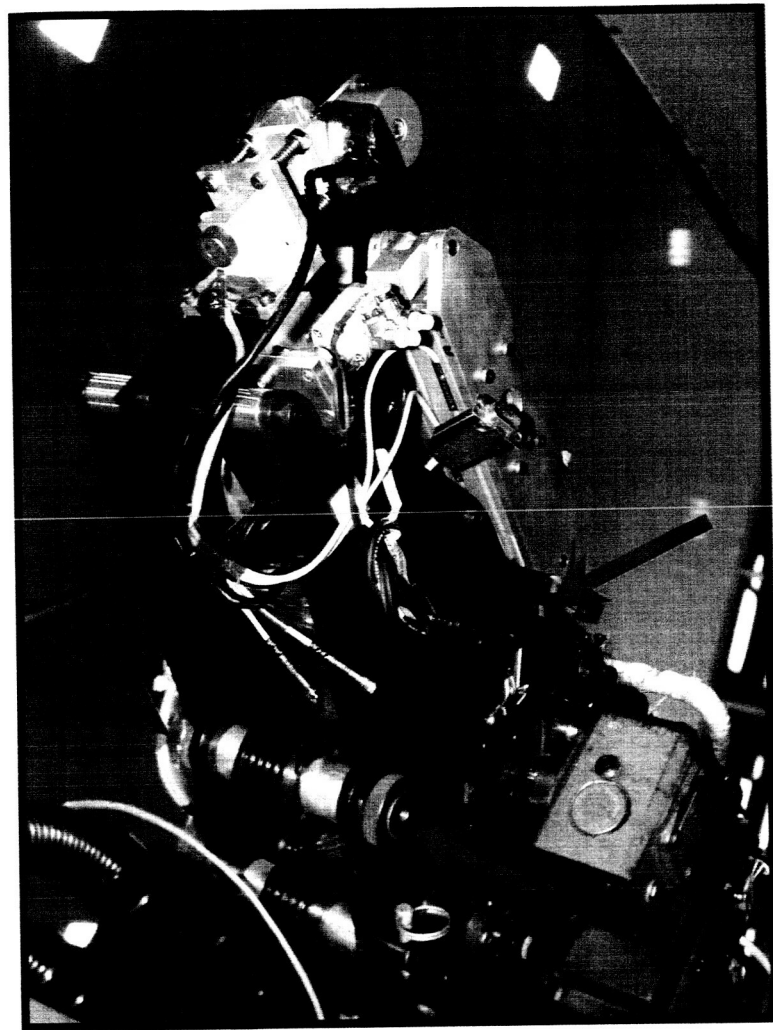
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## *Technical Details: Attach Points*



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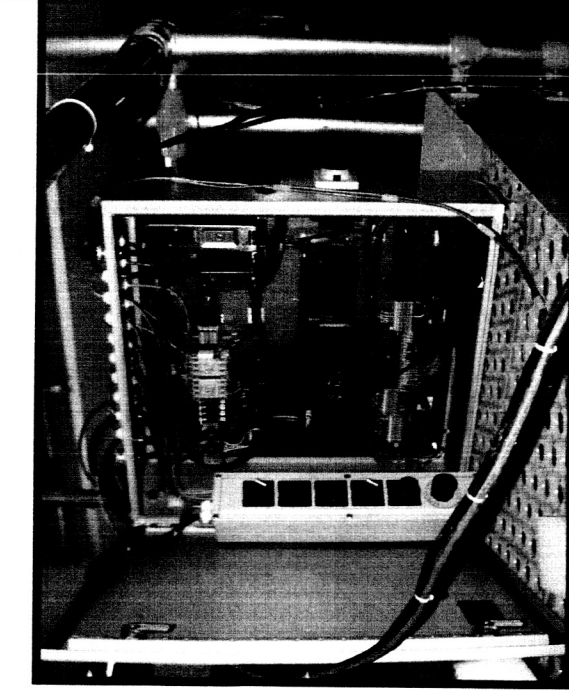
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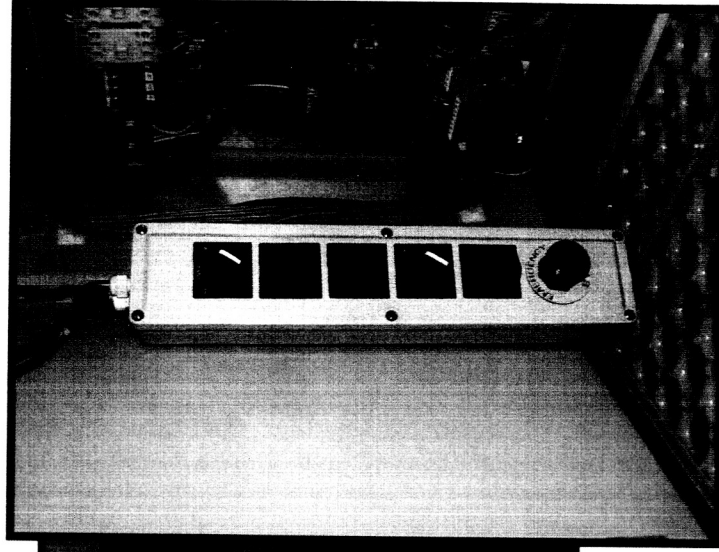
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## *Technical Details: Controls*



**Electrical Controls Enclosure**



**Hand Pendant Control**

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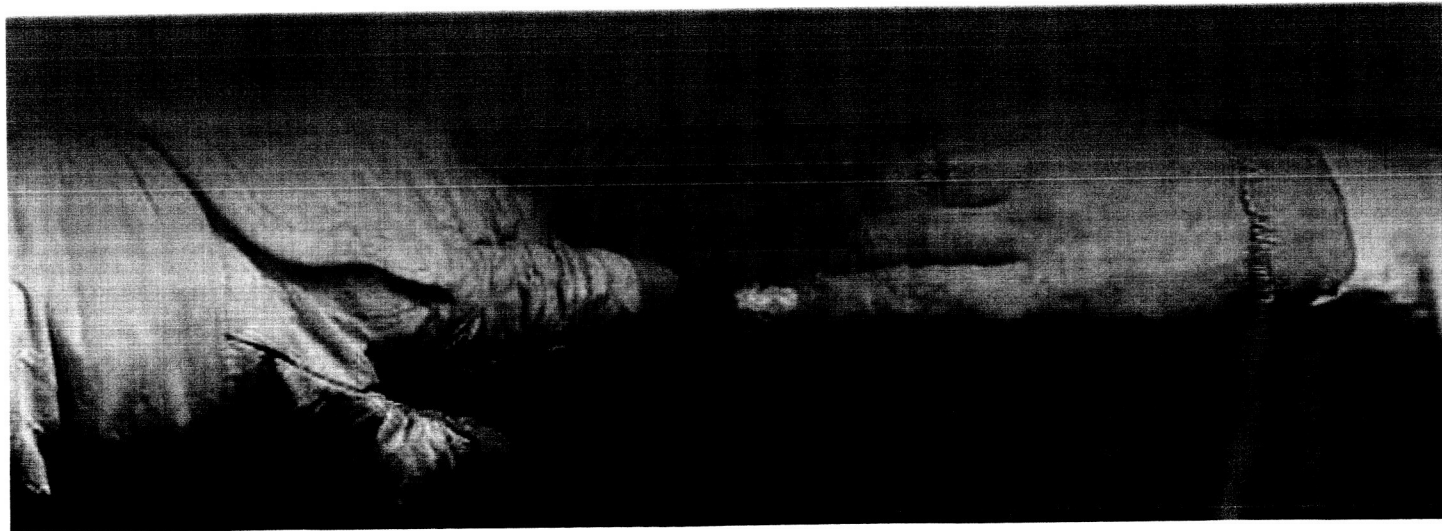
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#### *Technical Details: Prototype Specifics*

- PLC-Based Controls (ladder logic)
- Fully-Adjustable Device Sequencing and Timing via Speed/Motion Sensor
- Pneumatic Material Feed and Cut System uses Shop Air
- Electrical Power: 120 VAC
- Variable Power IR Radiant Heat Source for Substrate/Film Heating
- Simple Guillotine-Style Material Cutting System with Vacuum Hold
- Uses Host Machine Compaction Roller: May be Designed with Independent Heated Roller and/or Heated Trailing Shoe





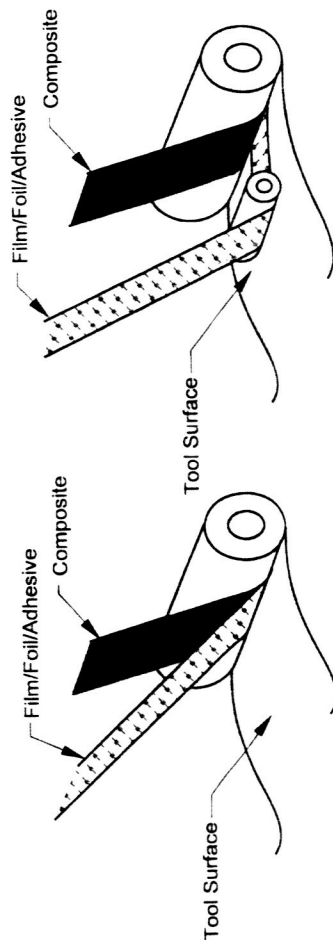
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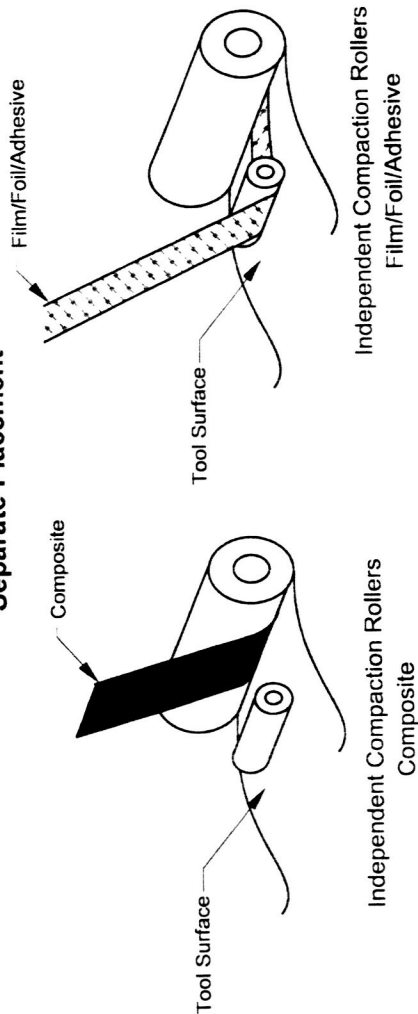
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#### Simultaneous Placement

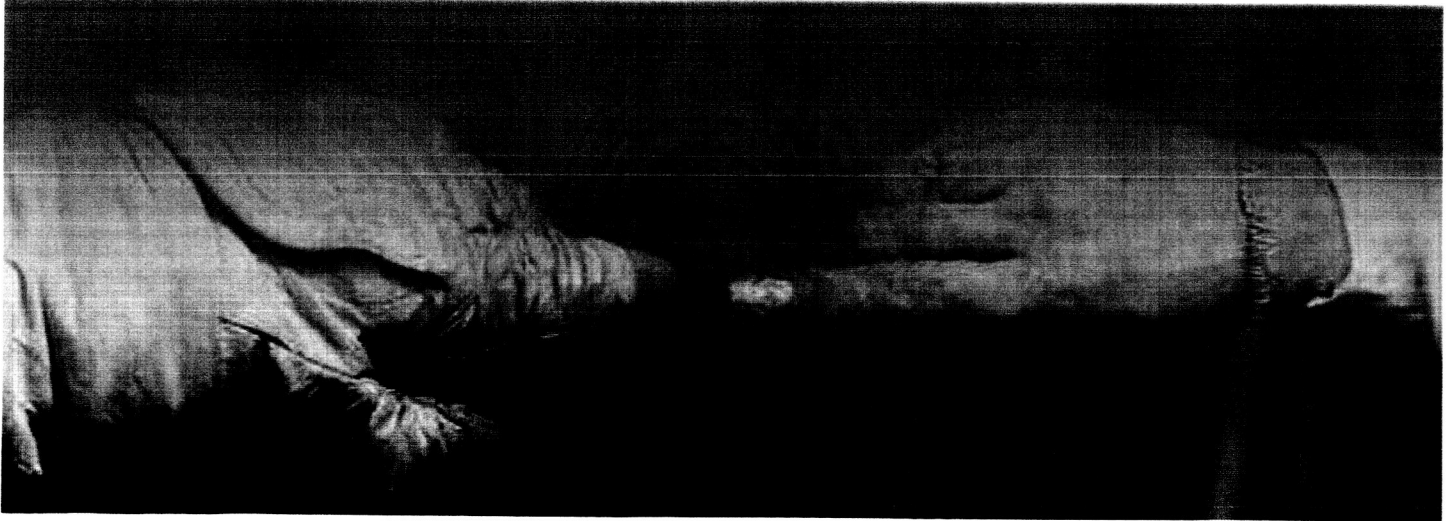


#### Separate Placement



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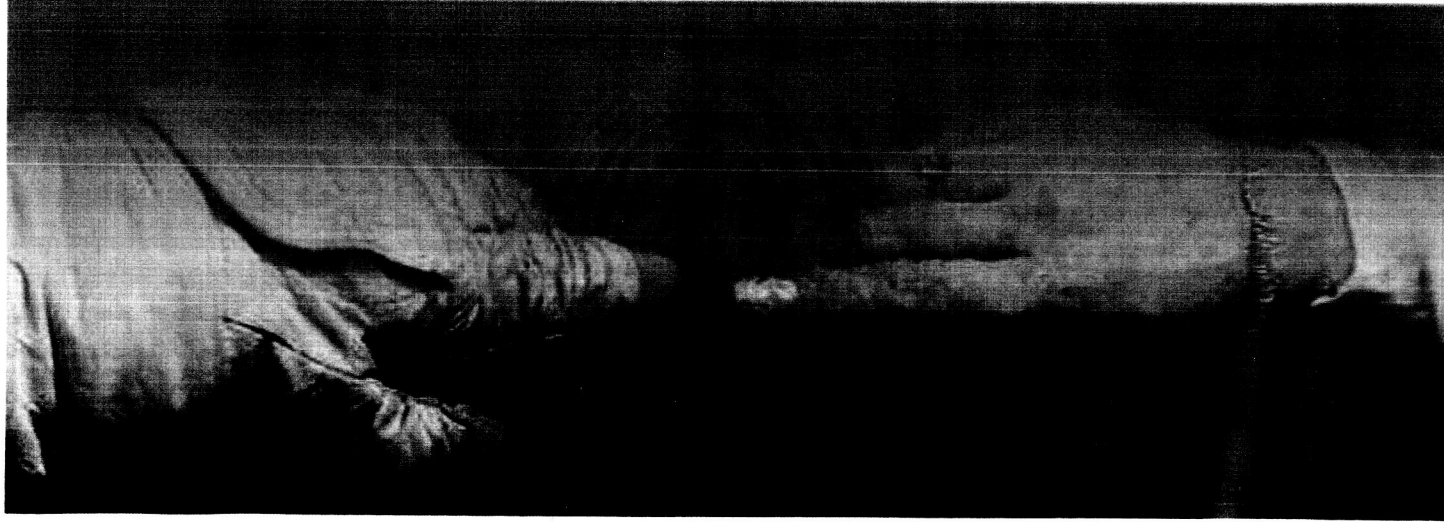
#### *Technical Advantages*

- Reduction or Elimination in Hand Lay-up
- Improvement in Quality
- Reduced Fabrication Costs
- Can Be Added to Existing FP Machines or Designed as Integral (fixed or detachable) for New Machines
- As an Add-On: Slave-Control for use of Existing Placement Files,

#### *Minimizing Integration Work*

- Quick Attach and Removal from Host Machine
- May Process T/S or T/P Films & Metallic Foils
- Capable of Processing Variable Material Thicknesses and Widths (1in., 3 in., etc.)





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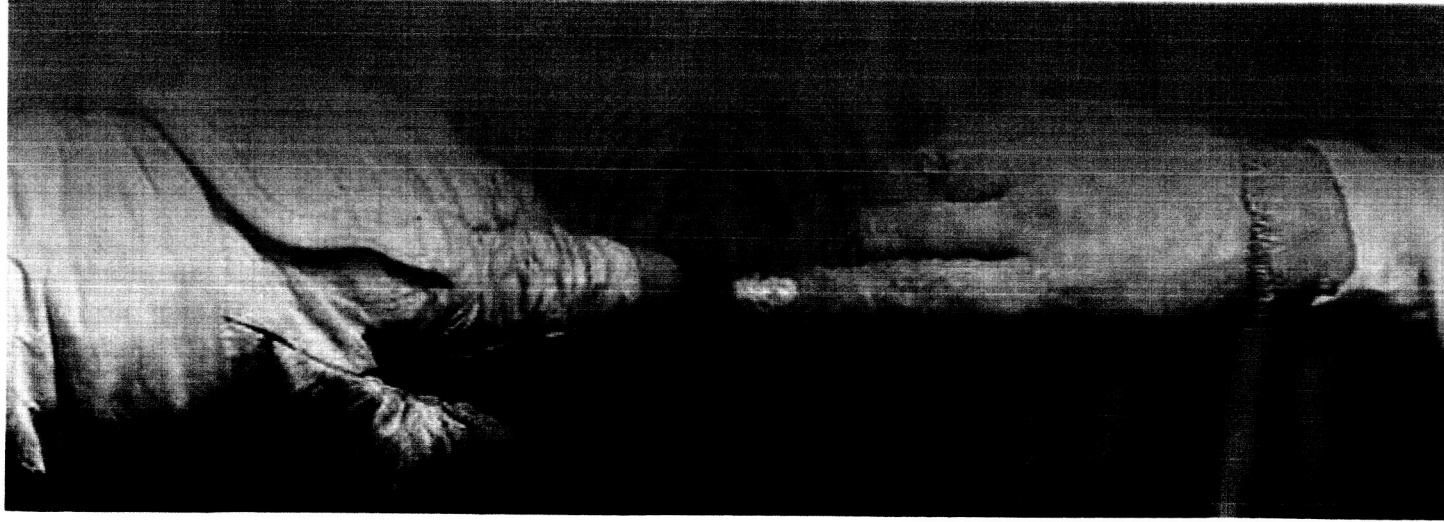
## *R&D Status*

Engineering Prototype Designed for Demonstrations  
And Testing at MSFC

Prototype has been Bench-Tested and Fit-Checked to Viper  
Placement Machine

Remaining Work:

Full-Scale Processing Demonstrations



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#### *Remaining Work: Initiate Processing Trials*

- Simultaneous and Separate Placement of Films and Composites
- Trials Using Materials of Different Thicknesses
- Trials with Metallic, Aluminized and Polymeric Films
- Overlap/Gap Studies
- Analysis and Mechanical Property Evaluations
- Publication of NASA Technical Memorandum



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## *NASA Plans/Options*

- Provisional Patent filed May '04
- Non-provisional Patent to be filed June '04
- NASA Seeks Partners to License this Technology

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## *NASA's Technology Transfer Program*

- Program seeks to stimulate commercial use of NASA-developed technology and infuse commercial technology into NASA missions.
- NASA is flexible in its agreements for licensing or partnerships for co-development. Opportunities in patent licensing include exclusive, non-exclusive, or exclusive field-of-use agreements.
- For more information, visit the NASA exhibit or contact Sammy Nabors: NASA/MSFC Technology Transfer, 256-544-5226, [sammy.nabors@nasa.gov](mailto:sammy.nabors@nasa.gov)



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## **Acknowledgements**

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**Peter Liao, Research Triangle Institute**

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